

# Combustion in Air-breathing Aero Engines

## Assignment No. 11

This assignment contains 8 multiple choice questions with 4 possible answers to each. Only one of the choice is correct and so select the choice that best answers the question. Correct choice rewards you with 1 point for each question. Wrong answers will reward you with 0 points (no negative marking). The questionnaire contains both numerical and concept-based questions. All the best!!!

1. Combustion achievable in afterburners cannot be completed in the gas turbine main combustor due to

- a) Limited availability of oxygen
- b) Limitation on turbine entry temperature
- c) Limitation on liner temperature
- d) Enhanced efficiency

Ans: b)

2. The Laser Induced Fluorescence (LIF) signal is proportional to

- a) pressure
- b) species mole fraction
- c) velocity
- d) laser wavelength

Ans b)

3. In LIF if the absorbed wavelength is given by  $\lambda_a$  and emitted wavelength is  $\lambda_e$  then

- a)  $\lambda_a > \lambda_e$
- b)  $\lambda_a < \lambda_e$
- c)  $\lambda_a = \lambda_e$
- d) None of the above

Ans: b)

4. A student is required to design a PIV experiment for measuring pipe flow with an average velocity of 100 m/s. The camera pixel size is 1mm and the window size is 32x32 px. Then the correct choice for time delay between two laser shots would be

- a) 8 micro seconds
- b) 18 micro seconds
- c) 80 micro seconds
- d) 8 milliseconds

Ans: c) Use  $0.1px < U\Delta t < \Delta s / 4$

5. In a bluff body stabilized flame, as blowoff is approached, Von Karmann instability could set in when

- a) pressure ratio between burnt and unburnt gases approaches unity
- b) velocity ratio between burnt and unburnt gases approaches unity
- c) density ratio between burnt and unburnt gases approaches unity
- d) density ratio between burnt and unburnt gases approaches zero

Ans c)

6. In afterburners the preferred mode of flame stabilization is using

- a) swirlers
- b) plasma
- c) bluff body
- d) pilot flame

Ans c)

7. The density ratio between unburnt to burnt gases in the gas turbine main combustor is given by  $R_m$  and that in the afterburner while it is operational is given by  $R_a$ .

a)  $R_m > R_a$

b)  $R_m = R_a < 2$

c)  $R_m < R_a$

d)  $R_m = R_a \geq 2$

Ans a). In an afterburner combustion happens in already vitiated and high temperature air.

8. Premixed flames can extinguish by stretch when

a)  $Le = 1$

b)  $0 < Le < 1$

c)  $Le > 1$

d)  $Le < 0$

Ans c)