

ME-662 CONVECTIVE HEAT AND MASS TRANSFER

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LECTURE-9 SIMILARITY SOLN TO TEMP BL - II

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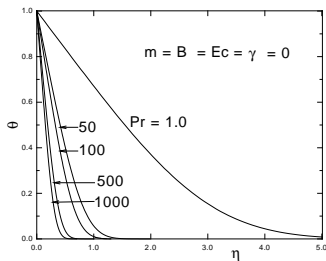
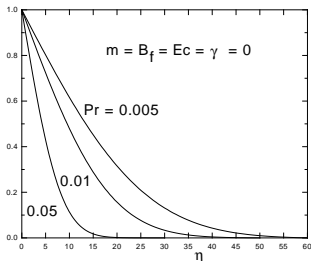
$$\theta'' + Pr \left[\left(\frac{m+1}{2} \right) f \theta' - \gamma f' \theta + 2 Ec (f'')^2 \right] = 0 \quad (1)$$

$$\theta(0) = 1 \text{ and } \theta(\infty) = 0$$

Similarity Solutions to study effect of

- 1 High and Low Prandtl Numbers (Pr)
- 2 Pressure Gradient (m)
- 3 Wall Temperature Variation (γ)
- 4 Viscous Dissipation (Ec)

Very Low and High Pr - L9($\frac{1}{12}$)



Liquid Metals

Pr	Δ^*	$-\theta'(0)$
0.005	53.3	0.04
0.01	39.3	0.0564
0.05	17.7	0.126

$$Nu_x = 0.564 (Re_x Pr)^{0.5}$$

Oils

Pr	Δ^*	$-\theta'(0)$
100	1.066	1.57
500	0.624	2.68
1000	0.495	3.28

$$Nu_x = 0.339 Re_x^{0.5} Pr^{0.33} \quad (\text{see later})$$

Effect of m - ($B_f = \gamma = Ec = 0$) - L9($\frac{2}{12}$)

$$\theta'' + Pr\left(\frac{m+1}{2}\right) f \theta' = 0 \quad \text{or} \quad \frac{d\theta'}{\theta'} = -Pr\left(\frac{m+1}{2}\right) f \quad (2)$$

Integration gives

$$\begin{aligned} \left[\ln(\theta') \right]_0^\eta &= -Pr\left(\frac{m+1}{2}\right) \int_0^\eta f d\eta \\ \theta' &= \theta'(0) \exp \left[-Pr\left(\frac{m+1}{2}\right) \int_0^\eta f d\eta \right] \\ \theta &= \theta'(0) \int_0^\eta \exp \left[-Pr\left(\frac{m+1}{2}\right) \int_0^\eta f d\eta \right] d\eta + C_1 \end{aligned}$$

But, $\theta(0) = 1$. Hence, $C_1 = 1$. Also, $\theta(\infty) = 0$. Hence,

$$\theta'(0) = - \left\{ \int_0^\infty \exp \left[-Pr\left(\frac{m+1}{2}\right) \int_0^\eta f d\eta \right] d\eta \right\}^{-1} \quad (3)$$

Effect of m - Moderate Pr) - $L9(\frac{3}{12})$

$(B_f = \gamma = Ec = 0)$ - H T Coef $h_x \propto x^{(\frac{m-1}{2})}$. For stagnation point, $h_x = \text{Const.}$

m	-0.085	-0.065	-0.04	0.0	0.33	1.0	4.0
	Pr=0.7						
$-\theta'(0)$	0.22	0.25	0.27	0.29	0.38	0.49	0.81
Δ^*	6.76	6.21	5.93	5.60	5.24	3.55	2.19
	Pr=5						
$-\theta'(0)$	0.40	0.47	0.52	0.57	0.79	1.03	1.71
Δ^*	3.67	3.21	2.98	2.73	2.07	1.59	0.98
	Pr=10						
$-\theta'(0)$	0.49	0.59	0.65	0.72	1.00	1.32	2.18
Δ^*	2.99	2.58	2.37	2.15	1.59	1.23	0.77
	Pr=25						
$-\theta'(0)$	0.64	0.79	0.88	0.98	1.37	1.81	3.10
Δ^*	2.31	1.95	1.76	1.59	1.19	0.92	0.56

Correlations: Effect of $m - Pr \gg 1$ L9($\frac{4}{12}$)

For $Pr \gg 1$, $\Delta \ll \delta$. Hence, $f'(\eta) \simeq f''(0, m) \eta$. And, $f(\eta) \simeq f''(0, m) \eta^2/2$. Then

$$\begin{aligned}\theta'(0) &= - \left\{ \int_0^\infty \exp \left[-Pr \left(\frac{m+1}{2} \right) \int_0^\eta f''(0, m) \frac{\eta^2}{2} d\eta \right] d\eta \right\}^{-1} \\ &= - \left[\int_0^\infty \exp \left\{ - \left(\frac{Pr (m+1) f''(0, m)}{12} \right) \eta^3 \right\} d\eta \right]^{-1}\end{aligned}$$

Using definition $\Gamma(n) = \int_0^\infty x^{n-1} e^{-x} dx$, it can be shown that

$$Nu_x Re_x^{-0.5} = -\theta'(0) = \left\{ \frac{Pr (m+1) f''(0, m)}{12} \right\}^{0.33} \times \frac{1}{\Gamma(4/3)}$$

where $\Gamma(4/3) \simeq 0.893$. Hence, for $m = 0$, with $f''(0) = 0.33$,
 $Nu_x \simeq 0.339 Re_x^{0.5} Pr^{0.33}$

Correlations: Effect of $m - Pr \ll 1$ L9($\frac{5}{12}$)

For $Pr \ll 1$, $\Delta \gg \delta$. Hence, $f'(\eta) \simeq 1.0$. And, $f(\eta) \simeq \eta$. Then

$$\begin{aligned}\theta'(0) &= - \left\{ \int_0^\infty \exp \left[-Pr \left(\frac{m+1}{2} \right) \int_0^\eta \eta \, d\eta \right] d\eta \right\}^{-1} \\ &= - \left[\int_0^\infty \exp \left\{ - \left(\frac{Pr(m+1)}{4} \right) \eta^2 \right\} d\eta \right]^{-1}\end{aligned}$$

Using definition $\text{erf}(x) = \frac{2}{\sqrt{\pi}} \int_0^x e^{-\eta^2} d\eta$ and $\text{erf}(\infty) = 1.0$

$$Nu_x Re_x^{-0.5} = -\theta'(0) = \sqrt{\frac{Pr(m+1)}{\pi}}$$

Hence, for $m = 0$, $Nu_x = 0.564 (Re_x Pr)^{0.5}$

Effect of $T_w - T_\infty = C x^\gamma$ - L9($\frac{6}{12}$)

$$m = B_f = Ec = 0, \quad \theta'' + Pr [0.5 f \theta' - \gamma f' \theta] = 0$$

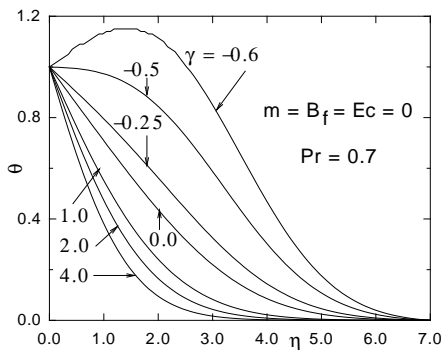
	Pr	0.7	5.0	10.0	25.0
γ					
4.0	$-\theta'(0)$	0.72	1.38	1.74	2.36
2.0	$-\theta'(0)$	0.582	1.12	1.41	1.91
1.0	$-\theta'(0)$	0.478	0.925	1.16	1.58
0.3	$-\theta'(0)$	0.366	0.713	0.898	1.22
0.0	$-\theta'(0)$	0.2913	0.572	0.721	0.976
-0.25	$-\theta'(0)$	0.195	0.388	0.489	0.662
-0.5	$-\theta'(0)$	0.0	0.0	0.0	0.0
-0.6	$-\theta'(0)$	-0.16	-0.45	-0.59	-0.84

Since $Ec = 0$, γ can take arbitrary values.

Special Case $\gamma < 0$ - L9($\frac{7}{12}$)

- 1 For $\gamma = -0.5$, temperature gradient at the wall is zero (**adiabatic case**) although $T_w > T_\infty$.
- 2 Since negative value of γ implies decreasing T_w with increasing x , the fluid particles close to the surface at a particular x arrive from upstream region of higher surface temperature.
- 3 These hotter fluid particles may thus **inhibit any heat transfer from the surface to the cooler free stream**.
- 4 By the same reasoning, **for $\gamma < -0.5$, $\theta > 1$** at some distance close to the surface and, **heat will flow into the surface** even if $T_w > T_\infty$.
Hence, **H T Coef < 0** (or, negative).

Effect of γ - Profiles - L9($\frac{8}{12}$)



- 1 Notice that at $\gamma = -0.6$, θ exceeds wall value $\theta(0) = 1$
- 2 At $\gamma = -0.5$, $\theta'(0) = 0$

Viscous Dissipation - L9($\frac{9}{12}$)

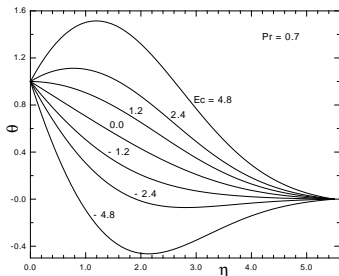
$$m = B_f = \gamma = 0 \text{ and } Pr = 0.7$$
$$\theta'' + 0.7 [0.5 f \theta' + 2 Ec (f'')^2] = 0$$

$$Ec = (U_\infty^2/2)/(Cp(T_w - T_\infty)) \quad \theta = (T - T_\infty)/(T_w - T_\infty)$$

Ec	-4.8	-2.4	-1.2	0.0	1.2	2.4	4.8
$-\theta'(0)$	1.458	0.875	0.583	0.292	0.004	-0.291	-0.874
Δ^*	0.861	2.70	4.84	5.26	5.35	5.39	5.43

- 1 $Ec < 0$ implies that $T_w < T_\infty$
- 2 $Ec > 0$ implies that $T_w > T_\infty$
- 3 For $Ec = 1.2$, $-\theta'(0) \simeq 0$ (**adiabatic case**)

Effect of Ec - Profiles - L9($\frac{9}{12}$)



- 1 For $Ec < -1.2$, $\theta < 0$ indicating that $T(\eta) > T_\infty$ within BL
- 2 For $Ec > 1.2$, $-\theta'(0) < 0$. Hence, $h_x < 0$ even when $T_w > T_\infty$
- 3 Both are effects of *local viscous heating* due to $\mu (\partial u / \partial y)^2$

Effect of B_f - L9($\frac{10}{12}$)

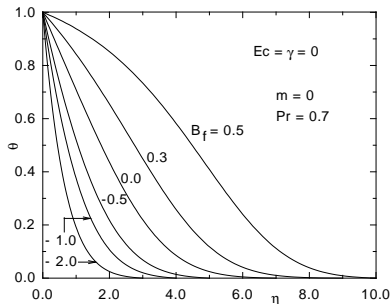
$$\gamma = Ec = 0, \quad \theta'' + 0.5 Pr (m + 1) f (m, B_f) \theta' = 0$$

	Pr	0.5	0.7	1.0	0.5	0.7	1.0
B_f		m=0			m=1		
-2.0	$-\theta' (0)$	1.12	1.52	2.10	1.22	1.62	2.20
-1.0	$-\theta' (0)$	0.672	0.872	1.17	0.799	1.012	1.32
-0.5	$-\theta' (0)$	0.459	0.570	0.726	0.606	0.738	0.917
0.0	$-\theta' (0)$	0.259	0.2913	0.330	0.434	0.493	0.664
0.3	$-\theta' (0)$	0.142	0.141	0.134	0.338	0.366	0.392
0.5	$-\theta' (0)$	0.064	0.051	0.035	0.281	0.292	0.293
1.0	$-\theta' (0)$	NA	NA	NA	0.163	0.145	0.116

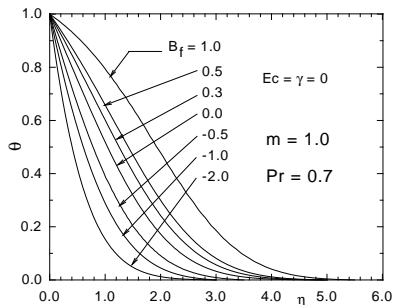
- 1 Only $0.5 < Pr < 1.0$ of interest (Gases)
- 2 Suction (negative B_f), increases H T by thinning BL
- 3 Blowing (positive B_f), decreases H T by thickening BL
- 4 For $m = 0$ and $B_f \geq 0.612$, Separation occurs

Effect of B_f - Profiles - L9($\frac{11}{12}$)

Flat Plate



Stagnation Point



Results for $Pr = 0.5$ and 1.0 are similar

Summary of Similarity Solns - L9($\frac{12}{12}$)

Velocity

- 1 Similarity Equations
$$f''' + \left(\frac{m+1}{2}\right) f f'' + m(1-f'^2) = 0$$
- 2 Boundary Conditions
$$f(0) = -B_f \left(\frac{2}{m+1}\right)$$
$$f'(0) = 0, f'(\infty) = 1$$
- 3 Similarity Conditions
$$U_\infty = C x^m$$
$$B_f = \left(\frac{V_w(x)}{U_\infty(x)}\right) Re_x^{0.5} = \text{Const}$$
$$\eta = y \times \sqrt{\frac{U_\infty}{\nu x}}$$
- 4 Similarity Solutions
$$C_{f,x}(m, B_f) = 2 f''(0) Re_x^{-0.5}$$

Temperature

- 1 Similarity Equations
$$\theta'' + Pr \left(\frac{m+1}{2}\right) f \theta' - Pr [\gamma f' \theta - 2 Ec (f'')^2] = 0$$
- 2 Boundary Conditions
$$\theta(0) = 1, \theta(\infty) = 0$$
- 3 Similarity Conditions
$$T_w(x) - T_\infty = \Delta T_{ref} x^\gamma$$
$$Ec_x = \frac{U_\infty^2(x)/2}{Cp(T_w(x) - T_\infty)} = \text{Const}$$

If $Ec \neq 0$, $\gamma = 2m$
- 4 Similarity Solutions
$$Nu_x Re_x^{-0.5}(m, B_f, Pr, \gamma, Ec) = -\theta'(0)$$