

Lecture – 39

# *Integrator Back-stepping; Linear Quadratic (LQ) Observer*

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# *Philosophy of Nonlinear Control Design Using Lyapunov Theory*

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# Philosophy of Feedback Control Design Using Lyapunov Theory

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Motivation :  $\dot{X} = f(X, U)$

Goal: Design  $U = \varphi(X)$  such that

$\dot{X} = f(X, \varphi(X))$  is asymptotically stable

Design Idea:

- \* Choose a pdf  $V_1(X)$
- \* Make  $\dot{V}_1(X) \leq -V_2(X)$ , where  $V_2(X) > 0$  (pdf)

# Feedback Control Design Using Lyapunov Theory: An Example

Problem: Design a stabilizing controller for the following system

$$\dot{x} = ax^2 - x^3 + u$$

Solution: Let  $V_1(X) = \frac{1}{2}x^2$

$$\begin{aligned}\dot{V}_1 &= x \dot{x} = x(ax^2 - x^3 + u) \\ &= ax^3 - x^4 + xu\end{aligned}$$

Let us choose  $V_2(X) = x^2$

$$\therefore \dot{V}_1(X) \leq -V_2(X)$$

$$\Rightarrow ax^3 - x^4 + xu \leq -x^2$$

# Philosophy of feedback control Design Using Lyapunov Theory

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$$xu \leq -x^2 + x^4 - ax^3$$

i.e. 
$$\boxed{u = -x + x^3 - ax^2}$$

Analysis: 
$$\dot{x} = ax^2 - x^3 - x + x^3 - ax^2$$

$$\dot{x} = -x$$

Advantage: The closed loop system is globally asymptotically stable.

Problem: The beneficial nonlinearity got cancelled.

(which is not desirable)

# Philosophy of feedback control Design Using Lyapunov Theory

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Let us choose:

$$V_2(X) = x^2 + x^4$$

Then

$$\dot{V}_1 \leq -V_2(X) \quad \text{leads to:}$$

$$ax^3 - x^4 + xu \leq -x^2 - x^4$$

$$ax^3 + xu \leq -x^2$$

$$ax^2 + u = -x \quad \text{or} \quad \boxed{u = -x - ax^2}$$

# Philosophy of feedback control Design Using Lyapunov Theory

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Closed Loop system:  $\dot{x} = ax^2 - x^3 - x - ax^2$

i.e.  $\dot{x} = -x^3 - x$

⇒ The destabilizing nonlinearity got cancelled,  
but the beneficial nonlinearity is retained !

Another Problem: If  $V_2(X) = x^2$ ,

$\dot{x} = -x$ , only if  $a$  is accurate

If the actual parameter value is  $\bar{a}$ , then the feedback  
loop operates with

# Philosophy of feedback control Design Using Lyapunov Theory

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$$\dot{x} = -x + \underbrace{(\bar{a} - a)}_{\text{Can be potentially destabilizing term if } (\bar{a} - a) \text{ is high}} x^2$$

Can be potentially  
destabilizing term  
if  $(\bar{a} - a)$  is high

*i.e.* The global stability reduces to local stability.

This excites robustness issues!

However, if  $V_2(X)$  is made "Sufficiently powerful",  
then the destabilizing effect can be minimized.

Hence, Lyapunov based designs can be "very robust"



# *Control Design Using Integrator Back-stepping*

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# Integrator Back-stepping

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**Problem :** Design a state feedback asymptotically stabilizing controller for the following system

$$\begin{cases} \dot{X} = f(X) + g(X)\xi \\ \dot{\xi} = u \end{cases}$$

where  $X \in \mathbb{R}^n$ ,  $\xi \in \mathbb{R}$ ,  $u \in \mathbb{R}$

Note:  $\begin{bmatrix} X \\ \xi \end{bmatrix} \in \mathbb{R}^{n+1}$ : State of the system,  $u$ : Control input (single input)

# Integrator Back-stepping

## Assumptions:

- \*  $f, g : D \rightarrow \mathbb{R}^n$  are smooth
- \*  $f(0) = 0$
- \* Considering state  $\xi$  as a "control input" of subsystem (1)  
we assume that  $\exists$  a state feedback control law  
of the form  $\xi = \varphi(X), \varphi(0) = 0$ . Moreover,  
 $\exists$  a Lyapunov function  $V_1 : D \rightarrow \mathbb{R}^+$  such that

$$\dot{V}_1(X) = \left( \frac{\partial V_1}{\partial X} \right)^T [f(X) + g(X)\varphi(X)] \leq -V_a(X), \quad \forall X \in D$$

where,  $V_a(X) : D \rightarrow \mathbb{R}^+$  is a pdf function.

# Integrator Back-stepping

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An important observation:

When  $X = 0$ ,  $\xi = \varphi(0) = 0$  &  $\dot{X} = f(0) = 0$

(i.e. everything is nice)

However, when  $\xi \rightarrow 0$ ,  $\dot{X} = f(X)$  and hence  $X \not\rightarrow 0$   
in general. That is the core problem!

$\therefore$  We need some algebraic manipulation as follows.

# Integrator Back-stepping

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Step - 1:

$$\begin{aligned}\dot{X} &= f(X) + g(X)\xi + g(X)\varphi(X) - g(X)\varphi(X) \\ &= f(X) + g(X)\varphi(X) + g(X)\underbrace{[\xi - \varphi(X)]}_z \\ &= f(X) + g(X)\varphi(X) + g(X)z\end{aligned}$$

By this construction, when  $z \rightarrow 0$ ,  $\dot{X} = f(X) + g(X)\varphi(X)$  which is asymptotically stable (i.e.  $X \rightarrow 0$ )!

# Integrator Back-stepping

$$\dot{z} = \dot{\xi} - \dot{\phi}$$

$$= \underbrace{u}_{v} - \dot{\phi}$$

[ This is backstepping, since  $\phi(X)$  is stepped back by differentiation ]

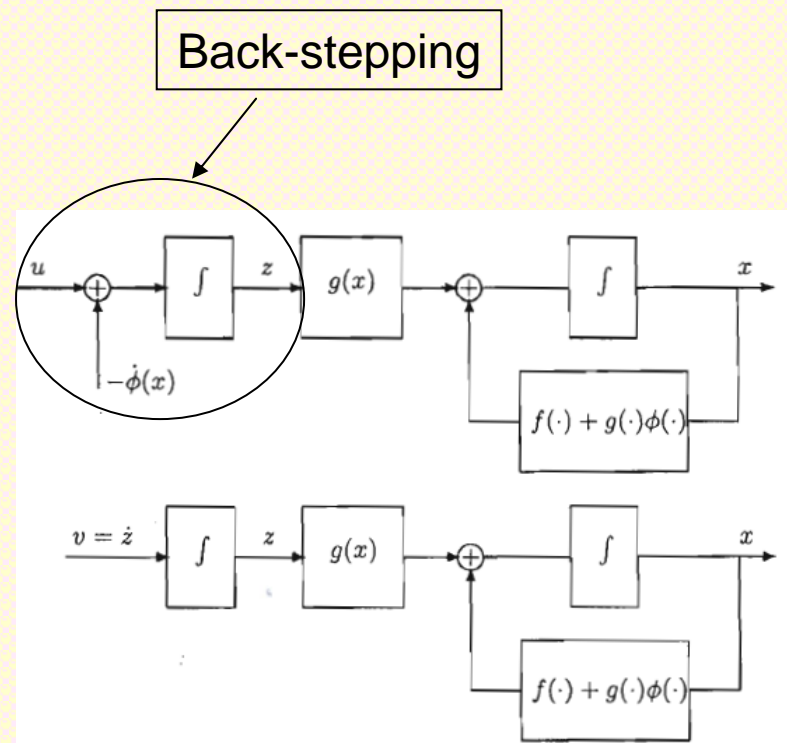
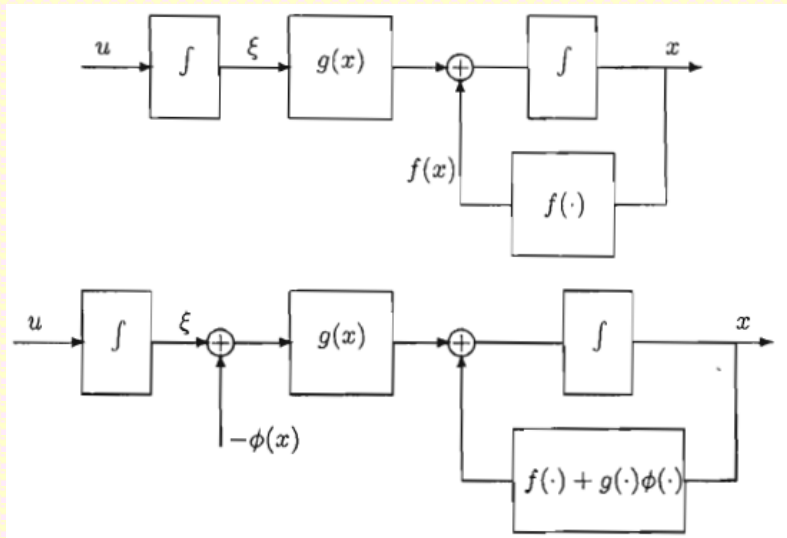
So, we have

$$\begin{aligned} \dot{X} &= f(X) + g(X)\phi(X) + g(X)z \\ \dot{z} &= v \end{aligned}$$

[ This system is equivalent to the original system ]

Note:  $\dot{\phi} = \left( \frac{\partial \phi}{\partial X} \right)^T \dot{X} = \left( \frac{\partial \phi}{\partial X} \right)^T [ f(X) + g(X)\xi ]$

# Back-stepping: Conceptual Block Diagram



**Ref:** H. J. Marquez, Nonlinear Control Systems: Analysis and Design, Wiley, 2003.

# Integrator Back-stepping

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Step-2: Let  $V(X, z) = V_1(X) + \frac{1}{2} z^2$

$$\begin{aligned} \text{Then } \dot{V} &= \underbrace{\left( \frac{\partial V_1}{\partial X} \right)^T [f(X) + g(X)\phi(X) + g(X)z]}_{\leq -V_a(X)} + z v \\ &\leq -V_a(X) + \left[ \left( \frac{\partial V_1}{\partial X} \right)^T g(X) + v \right] z \end{aligned}$$

$$\text{Let } v = - \left( \frac{\partial V_1}{\partial X} \right)^T g(X) - k z, \quad k > 0$$

$$\text{Then } \dot{V} \leq \underbrace{-V_a(X)}_{\text{ndf}} \underbrace{-k z^2}_{\text{nddf}} < 0 \quad (\text{ndf})$$



# Integrator Back-stepping

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Control Solution:

$$v = u - \dot{\phi} = - \left( \frac{\partial V_1}{\partial X} \right)^T g(X) - k z$$

$$u = \dot{\phi} - \left( \frac{\partial V_1}{\partial X} \right)^T g(X) - k [\xi - \phi(X)]$$

$$\text{where, } \dot{\phi} = \left( \frac{\partial \phi}{\partial X} \right)^T [f(X) + g(X)\xi], k > 0$$

[ Note: In the design, there is a need to design  $\phi(X)$  first ]

# Integrator Back-stepping: An Example

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Problem:  $\dot{x}_1 = ax_1^2 - x_1^3 + x_2$   
 $\dot{x}_2 = u$

Solution :

$$X = x_1, f(x_1) = ax_1^2 - x_1^3, \xi = x_2, g(x_1) = 1$$

To find  $\varphi(x_1)$ :

$$V_1(x_1) = \frac{1}{2} x_1^2$$

$$\dot{V}_1 = x_1 \dot{x}_1 = x_1 (ax_1^2 - x_1^3 + x_2) \leq -\underbrace{V_a(x_1)}_{x_1^2 + x_1^4}$$

# Integrator Back-stepping: An Example

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$$ax_1^3 - \cancel{x_1^4} + x_1x_2 \leq -x_1^2 - \cancel{x_1^4}$$

$$x_1(ax_1^2 + x_2) \leq -x_1^2$$

$$\text{Let } ax_1^2 + x_2 = -x_1$$

$$\Rightarrow x_2 = (-ax_1^2 - x_1) \triangleq \varphi(x_1)$$

$$\text{Modified system: } \dot{x}_1 = ax_1^2 - x_1^3 + \varphi(x_1) + \overbrace{[x_2 - \varphi(x_1)]}^z$$

$$\dot{z} = v \triangleq (\cancel{\dot{x}_2}^u - \dot{\varphi}(x_1))$$

$$\text{Let } V(x_1, z) = V_1(x_1) + \frac{1}{2}z^2$$

$$\dot{V} = \dot{V}_1 + z v = \left( \frac{\partial V_1}{\partial x_1} \right) [ax_1^2 - x_1^3 + \varphi(x_1)] + \left( \frac{\partial V_1}{\partial x_1} + v \right) z$$

# Integrator Back-stepping: An Example

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$$\text{Let } v = -\left(\frac{\partial V_1}{\partial x_1}\right) - k z, \quad k > 0$$

$$u - \dot{\varphi} = -x_1 - k [x_2 - \varphi(x_1)]$$

$$u = \frac{\partial \varphi}{\partial x_1} (ax_1^2 - x_1^3 + x_2) - x_1 - k [x_2 - (-ax_1^2 - x_1)]$$

$$= (-2ax_1 - 1)(ax_1^2 - x_1^3 + x_2) - x_1 - k [x_2 + ax_1^2 + x_1]$$

$$u = -(1 + 2ax_1)(ax_1^2 - x_1^3 + x_2) - x_1 - k(x_1 + x_2 + ax_1^2)$$

where  $k > 0$

# Integrator Back-stepping: An Example

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Note: The composite Lyapunov function is:

$$\begin{aligned}V &= V_1 + \frac{1}{2} z^2 \\ &= \frac{1}{2} x_1^2 + \frac{1}{2} [x_2 - \varphi(x_1)]^2 \\ &= \frac{1}{2} x_1^2 + \frac{1}{2} (x_2 + x_1 + ax_1^2)^2\end{aligned}$$

# Integrator Back-stepping: More General Case

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System Dynamics:

$$\dot{X} = f(X) + g(X)\xi_1$$

$$\dot{\xi}_1 = \xi_2$$

$$\dot{\xi}_2 = u$$

Idea : Successive iteration.

[ Note: The procedure for  $n^{th}$  order system is entirely analogous ]

# Integrator Back-stepping: More General Case

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Step-1: Consider the subsystem

$$\dot{X} = f(X) + g(X)\xi_1$$

$$\dot{\xi}_1 = \xi_2$$

Assumption:

$\xi_1 = \varphi(X)$  is a stabilizing feedback law for

$$\dot{X} = f(X) + g(X)\xi_1$$

and  $V_1(X)$  is the corresponding Lyapunov function.

# Integrator Back-stepping: More General Case

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By the result obtained before, we have

$$\xi_2 = \left( \frac{\partial \varphi(X)}{\partial X} \right)^T \underbrace{[f(X) + g(X)\xi_1]}_{\dot{X}} - \left( \frac{\partial V_1}{\partial X} \right)^T g(X) - k[\xi_1 - \varphi(X)]$$
$$\triangleq \varphi_1(X, \xi_1) \quad (k > 0)$$

We also have  $V_2 = V_1 + \frac{1}{2}[\xi_1 - \varphi(X)]^2$



# Integrator Back-stepping: More General Case

Step - 2:

$$\dot{X}_1 \triangleq \begin{bmatrix} \dot{X} \\ \dot{\xi}_1 \end{bmatrix} = \underbrace{\begin{bmatrix} f(X) + g(X)\xi_1 \\ 0 \end{bmatrix}}_{f_1(X_1)} + \underbrace{\begin{bmatrix} 0 \\ 1 \end{bmatrix}}_{g_1(X_1)} \xi_2$$

$$\dot{\xi}_2 = u \quad \text{where, } X_1 \triangleq \begin{bmatrix} X \\ \xi_1 \end{bmatrix}$$

$\therefore$  Using the same idea,

$$u = \left( \frac{\partial \varphi_1}{\partial X_1} \right)^T [f_1(X_1) + g_1(X_1)\xi_2] - \left( \frac{\partial V_2}{\partial X_1} \right)^T g_1(X_1) - k_1 [\xi_2 - \varphi_1(X_1)], \quad k_1 > 0$$

$$\text{and } V = V_2 + \frac{1}{2} [\xi_2 - \varphi_1(X_1)]^2 = V_1 + \frac{1}{2} [\xi_1 - \varphi(X)]^2 + \frac{1}{2} [\xi_2 - \varphi_1(X_1)]^2$$

# Integrator Back-stepping for Strict Feedback Systems

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## System Dynamics

$$\begin{aligned}\dot{X} &= f(X) + g(X)\xi_1 \\ \dot{\xi}_1 &= f_1(X, \xi_1) + g_1(X, \xi_1)\xi_2 \\ \dot{\xi}_2 &= f_2(X, \xi_1, \xi_2) + g_2(X, \xi_1, \xi_2)\xi_3 \\ &\vdots \\ &\vdots \\ \dot{\xi}_k &= f_k(X, \xi_1, \dots, \xi_k) + g_k(X, \xi_1, \dots, \xi_k)u\end{aligned}$$

## Strong Assumption:

$$g_1(X, \xi_1), g_2(X, \xi_1, \xi_2), \dots, g_k(X, \xi_1, \dots, \xi_k) \neq 0$$

over the domain of interest  $\forall t$

# Integrator Back-stepping for Strict Feedback Systems

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Special Case:  $\dot{X} = f(X) + g(X)\xi$   
 $\dot{\xi} = f_a(X, \xi) + g_a(X, \xi)u$

Solution:

Define  $\dot{\xi} = v$  and carryout the design for  $v$  as before.

Finally  $f_a(X, \xi) + g_a(X, \xi)u = v$

*i.e.* 
$$u = \frac{1}{g_a(X, \xi)} [v - f_a(X, \xi)]$$

Note: By assumption,  $g_a(X, \xi) \neq 0 \quad \forall t$

# *Linear Quadratic (LQ) Observer*

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# Why Observers?

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- State feedback control designs need the state information for control computation
- In practice all the state variables are not available for feedback. Possible reasons are:
  - Non-availability of sensors
  - Expensive sensors
  - Quality of some sensors may not be acceptable due to noise (its an issue in output feedback control design as well)
- A state observer estimates the state variables based on the measurement of some of the output variables as well as the plant information.

# Observer

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- An observer is a dynamic system whose output is an estimate of the state vector  $X$ 
  - Full-order Observer
  - Reduced-order Observer
- **Observability** condition must be satisfied for designing an observer (this is true for filter design as well)

# Observer Design for Linear Systems

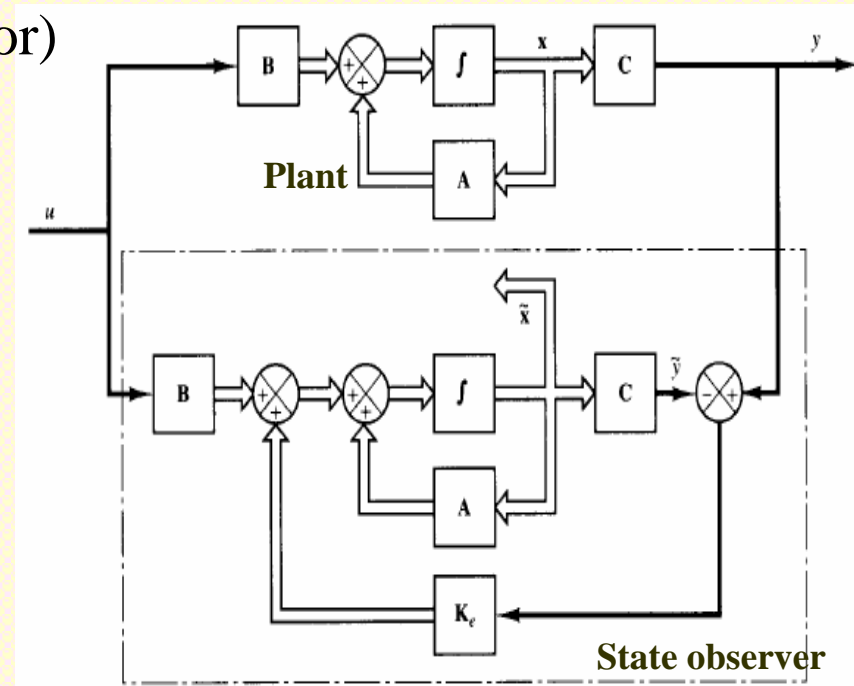
**Plant :**  $\dot{X} = AX + BU$

$Y = CX$  (sensor output vector)

Let the observed state be  $\hat{X}$  and the  
**Observer dynamics** be

$$\dot{\hat{X}} = \hat{A}\hat{X} + \hat{B}U + K_e Y$$

**Error :**  $\tilde{X} \triangleq (X - \hat{X})$



# Observer Design for Linear Systems

**Error Dynamics:**  $\dot{\tilde{X}} = \dot{X} - \dot{\hat{X}}$

$$= (AX + BU) - (\hat{A}\hat{X} + \hat{B}U + K_e Y)$$

Add and Subtract  $\tilde{A}X$  and substitute  $Y = CX$

$$\begin{aligned}\dot{\tilde{X}} &= AX - \hat{A}X + \hat{A}X - \hat{A}\hat{X} + BU - \hat{B}U - K_e C X \\ &= (A - \hat{A})X + \hat{A}(X - \hat{X}) + (B - \hat{B})U - K_e C X \\ &= \hat{A}\tilde{X} + (A - \hat{A} - K_e C)X + (B - \hat{B})U\end{aligned}$$

- Goals:**
1. Make the error dynamics independent of  $X$   
( $\because X$  may be large, even though  $\tilde{X}$  may be small)
  2. Eliminate the effect of  $U$  from error dynamics



# Observer Design for Linear Systems

This can be done by enforcing  $A - \hat{A} - K_e C = 0$   
and  $B - \hat{B} = 0$

Necessary and sufficient condition  
for the existence of  $K_e$ :

The system should be “observable”.

This results in

$$\hat{A} = A - K_e C$$

$$\hat{B} = B$$

**Observer dynamics:**

$$\dot{\hat{X}} = A\hat{X} + BU + K_e (Y - C\hat{X})$$

# Observer Design: Full Order

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- Order of the observer is same as that of the system (i.e. all states are estimated, irrespective of whether they are measured or not).
- **Goal:** Obtain gain  $\mathbf{K}_e$  such that the error dynamics are asymptotically stable with sufficient speed of response.

This means that  $\hat{\mathbf{A}} = \mathbf{A} - \mathbf{K}_e \mathbf{C}$  is Hurwitz (i.e. it has all eigenvalues strictly in the left half plane).

- Note:  $\hat{\mathbf{A}}^T = \mathbf{A}^T - \mathbf{C}^T \mathbf{K}_e^T$  and the eigen values of both  $\hat{\mathbf{A}}$  and  $\hat{\mathbf{A}}^T$  are **same!**

# Comparison of Control and Observer Design Philosophies

## Control Design

- CL Dynamics

$$\dot{X} = (A - BK)X$$

- Objective

$$X(t) \rightarrow 0, \text{ as } t \rightarrow \infty$$

## Observer Design

- CL Error Dynamics

$$\dot{\tilde{X}} = \hat{A}\tilde{X} = (A - K_e C)\tilde{X}$$

- Objective

$$\tilde{X}(t) \rightarrow 0, \text{ as } t \rightarrow \infty$$

- Notice that

$$\begin{aligned} \lambda(A - K_e C) &= \lambda \left[ (A - K_e C)^T \right] \\ &= \lambda \left( A^T - C^T K_e^T \right) \end{aligned}$$

# Algebraic Riccati Equation (ARE) Based Observer Design

## System

$$\dot{X} = AX + BU$$

$$Y = CX$$

$$M = \left[ B \mid AB \mid \cdots \mid A^{n-1}B \right]$$

$$N = \left[ C^T \mid A^T C^T \mid \cdots \mid A^{T^{n-1}} C^T \right]$$

## LQR Design

$$U = -KX$$

## Dual System

$$\dot{Z} = A^T Z + C^T V$$

$$n = B^T Z$$

$$M = \left[ C^T \mid A^T C^T \mid \cdots \mid A^{T^{n-1}} C^T \right]$$

$$N = \left[ B \mid AB \mid \cdots \mid A^{n-1}B \right]$$

# ARE Based Observer Design

## CL system (control design)

$$\dot{X} = (A - BK)X$$

$$X \rightarrow 0 \text{ as } t \rightarrow \infty$$

$$K = R^{-1}B^T P, \quad P > 0$$

where,

$$PA + A^T P - PBR^{-1}B^T P + Q = 0$$

## Error Dynamics

$$\dot{\tilde{X}} = (A - K_e C) \tilde{X}$$

$$(A - K_e C)^T = A^T - C^T K_e^T$$

## Analogous

$$K_e^T = R^{-1}CP$$

where,

$$PA^T + AP - PC^T R^{-1}CP + Q = 0$$

## Observer Dynamics

$$\dot{\hat{X}} = A\hat{X} + BU + K_e(Y - C\hat{X})$$

Acts like a controller gain

# *Continuous-time Kalman Filter Design for Linear Time Invariant (LTI) Systems*

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# Problem Statement

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System Dynamics:  $\dot{X} = AX + BU + GW$      $W(t)$ : Process noise vector

Measured Output:  $Y = CX + V$      $V(t)$ : Sensor noise vector

Assumptions:

(i)  $X(0) \sim (\tilde{X}_0, P_0)$ ,  $W(t) \sim (0, Q)$  and  $V(t) \sim (0, R)$

are "mutually orthogonal" [ $X(0)$ : initial condition for  $X$ ]

(ii)  $W(t)$  and  $V(t)$  are uncorrelated white noise

(iii)  $E[W(t) W^T(t + \tau)] = Q \delta(\tau)$ ,  $Q \geq 0$  (psdf)

$E[V(t) V^T(t + \tau)] = R \delta(\tau)$ ,  $R > 0$  (pdf)

# Problem Statement

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## **Objective:**

To obtain an estimate of the state vector  $\hat{X}(t)$  using the state dynamics as well as a "sequence of measurements" as accurate as possible.

i.e., to make sure that the error  $\tilde{X}(t) \triangleq [X(t) - \hat{X}(t)]$  becomes very small (ideally  $\tilde{X}(t) \rightarrow 0$ ) as  $t \rightarrow \infty$ .



# Observer/Estimator/Filter Dynamics

$$\dot{\hat{X}} = A\hat{X} + BU + K_e (Y - \hat{Y})$$

- where
- (i)  $\hat{X} = E(X)$  : Estimate of the state  $X$
  - (ii)  $\hat{Y} = E(Y)$  : Estimate of the output  $Y$   
 $= E(CX + V)$   
 $= E(CX) + E(V)$   
 $= CE(X) \quad (\because E(V) = 0)$   
 $= C\hat{X}$
  - (iii)  $K_e$  : Estimator/Filter/Kalman Gain

**Problem :** How to design  $K_e$  ?

## Solution: Summary

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- (i) Initialize  $\hat{X}(0)$
- (ii) Solve for Riccati matrix  $P$  from the Filter ARE:

$$AP + PA^T - PC^T R^{-1} CP + GQG^T = 0$$

- (iii) Compute Kalman Gain:

$$K_e = PC^T R^{-1}$$

- (iv) Propagate the Filter dynamics:

$$\dot{\hat{X}} = A\hat{X} + BU + K_e(Y - C\hat{X})$$

where  $Y$  is the measurement vector

**Thanks for the Attention...!**

