GATE 2014: General Instructions during Examination

- 1. Total duration of the GATE examination is **180** minutes.
- 2. The clock will be set at the server. The countdown timer at the top right conscreen will display the remaining time available for you to complete the examinate the timer reaches zero, the examination will end by itself. You need terminate the examination or submit your paper.
- 3. Any useful data required for your paper can be viewed by clicking on the **Common Data** button that appears on the screen.
- 4. Use the scribble pad provided to you for any rough work. Submit the scribl the end of the examination.
- 5. You are allowed to use a non-programmable type calculator, however, sha calculators is not allowed.
- 6. The Question Palette displayed on the right side of screen will show the state each question using one of the following symbols:

1 You have not visited the question yet.

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AE15M027

Aerospace Engineering IITMadras

You have not answered the question.

You have answered the question.

You have NOT answered the question, but have marked the question for review.

You have answered the question, but marked it for review.

The **Marked for Review** status for a question simply indicates that you would that question again. *If a question is answered, but marked for review, then the be considered for evaluation unless the status is modified by the candidate.*

Navigating to a Question:

- 7. To answer a question, do the following:
 - a. Click on the question number in the Question Palette to go to directly.
 - b. Select an answer for a multiple choice type question by clicking on the placed before the 4 choices, namely A, B, C and D. Use the v keypad to enter a number as answer for a numerical type question.
 - c. Click on **Save & Next** to save your answer for the current question a to the next question.
 - d. Click on **Mark for Review & Next** to save your answer for the curre and also mark it for review, and then go to the next question.

Caution: Note that your answer for the current question will not be saved, if you to another question directly by clicking on a question number without saving the previous question.

You can view all the questions by clicking on the **Question Paper** button provided, so that if you want you can just see the entire question paper at a glassian paper

Answering a Question:

- 8. Procedure for answering a multiple choice (MCQ) type question:
 - a. Choose one answer from the 4 options (A,B,C,D) given below the click on the bubble placed before the chosen option.
 - b. To deselect your chosen answer, click on the bubble of the chosen op or click on the **Clear Response** button.
 - c. To change your chosen answer, click on the bubble of another option
 - d. To save your answer, you MUST click on the Save & Next button.
- 9. Procedure for answering a numerical answer type question:
 - a. To enter a number as your answer, use the virtual numerical keypad.
 - b. A fraction (e.g. -0.3 or -.3) can be entered as an answer with before the decimal point. As many as four decimal points, e.g. 0.003 or -932.6711 or 12.82 can be entered.
 - c. To clear your answer, click on the **Clear Response** button.
 - d. To save your answer, you MUST click on the Save & Next button
- 10. To mark a question for review, click on the Mark for Review & Next but answer is selected (for MCQ) or entered (for numerical answer type) that is Marked for Review, that answer will be considered in the extra the status is modified by the candidate.
- 11.To change your answer to a question that has already been answer that question for answering and then follow the procedure for answering t question.
- 12. Note that ONLY Questions for which answers are **saved** or **marked for realization**.

Choosing a Section:

- 13. Sections in this question paper are displayed on the top bar of the screen in a Section can be viewed by clicking on the name of that Section. The Seare currently viewing will be highlighted.
- 14.A checkbox is displayed for every optional Section, if any, in the Question select the optional Section for answering, click on the checkbox for that Section for answering, click on the checkbox for that Section for answering is a select the optional Section for answering.
- 15.If the checkbox for an optional Section is not selected, the **Save & Next** the **Mark for Review & Next** button will NOT be enabled for that Section

- only be able to see questions in this Section, but you will not be a questions in the Section.
- 16.After clicking the **Save & Next** button for the last question in a Sautomatically be taken to the first question of the next Section in sequence
- 17. You can move the mouse cursor over the name of a Section to view the are status for that Section.

Changing the Optional Section:

- 18. After answering the chosen optional Section, partially or completely, you the optional Section by selecting the checkbox for a new Section th attempt. A warning message will appear along with a table showing the nuquestions in the previous of the previous of the previous of the sections. Click on a checkbox against a Section will be previous of the previous of
- 19. If you deselect the checkbox for an optional Section in the top be warning message will appear: "Deselecting the checkbox will DELETE all the for questions in this Section. Do you want to deselect this Section?" If you deselect, click on the **RESET** button. If you do not want to deselect, click of button.
- 20. You can shuffle between different Sections or change the optional number of times.

GATE 2014 Examination

AE: Aerospace Engineering

Duration: 180 minutes Maximum Marks: 100

Read the following instructions carefully.

- 1. To login, enter your Registration Number and password provided to you. Kindly go through the various symbols used in the test and understand their meaning before you start the examination.
- 2. Once you login and after the start of the examination, you can view all the questions in the question paper, by clicking on the **View All Questions** in the screen.
- 3. This question paper consists of **2 sections** neral Aptitude (GA) for **15 marks** the subject specific GATE paper for **85 marks** oth these sections are compulsory.

 The GA section consists of **10** mestions. Question numbers 1 to 5 are of 1-mark each, while question.
 - The GA section consists of **10**questions. Question numbers 1 to 5 are of 1-mark each, while question numbers 6 to 10 are of 2-mark each.
 - The subject specific GATE paper section consists of **55**questions, out of which question numbers 1 to 25 are of 1-mark each, while question numbers 26 to 55 are of 2-mark each.
- 4. Depending upon the GATE paper, there may be useful common data that may be required for answering the questions. If the paper has such useful data, the same can be viewed by clicking on the **Useful Common Data** that appears at the top, right hand side of the screen.
- 5. The computer allotted to you at the examination center runs specialized software that permits only one answer to be selected for multiple-choice questions using a mouse and to enter a suitable number for the numerical answer type questions using the virtual keyboard and mouse.
- 6. Your answers shall be updated and saved on a server periodically and also at the end of the examination. The examination will **stop automatigatly** end of **180 minutes**
- 7. In each paper a candidate can answer a total of 65 questions carrying 100 marks.
- 8. The question paper may consist of questions of **multiple choice type**) and **numerical answer** type
- 9. Multiple choice type questions will have four choices against A, B, C, D, out of which only **ONE** is the correct answer. The candidate has to choose the correct answer by clicking on the bubble () placed before the choice.
- 10. For numerical answer type questions, each question will have a numerical answer and there will not be any choices. **For these questions, the answer should**us**bg** t**entiered**keyboard that appears on the monitor and the mouse.
- 11. All questions that are not attempted will result in zero marks. However, wrong answers for multiple choice type questions (MCQ) will result in **NEGATIVE** marks. For all MCQ questions a wrong answer will result in deduction of ¹/₂ marks for a 1-mark question and ²/₂ marks for a 2-mark question.
- 12. There is **NO NEGATIVEMARKING** for questions of **NUMERICALANSWER TYPE**
- 13. Non-programmable type Calculator is allowed. Charts, graph sheets, and mathematical tables are **NOT** allowed in the Examination Hall. You must use the Scribble pad provided to you at the examination centre for all your rough work. The Scribble Pad has to be returned at the end of the examination.

Declaration by the candidate:

"I have read and understood all the above instructions. I have also read and understood clearly the instructions given on the admit card and shall follow the same. I also understand that in case I am found to violate any of these instructions, my candidature is liable to be cancelled. I also confirm that at the start of the examination all the computer hardware allotted to me are in proper working condition".

Q. 1 - Q. 5 carry one mark each.

Q.1 A student is required to demonstrate a high level of <u>comprehension</u> of the subject, especially in the social sciences.

The word closest in meaning to comprehension is

- (A) understanding
- (B) meaning
- (C) concentration
- (D) stability

Genera

Q.2 Choose the most appropriate word from the options given below to complete the following sentence.

One of his biggest _____ was his ability to forgive.

- (A) vice
- (B) virtues
- (C) choices
- (D) stre gu

Q.3 Rajan was not happy that Sajan decided to do the project on his own. In other lines with a splain was not happy that Sajan decided to do the project on his own. In other lines with the project on his own. In other lines with the project on his own.

Which one of the statements below is logically valid and can be inferred from the above sentences?

- (A) Rajan has decided to work only in a group.
- (B) Rajan and Sajan were formed into a group against their wishes.
- (C) Sajan had decided to give in to Rajan's request to cork with him.
- (D) Rajan had believed that Sajan and he would have working to be ber.
- Q.4 If $y=5x^2+3$, then the tangent at x=0, y=3
 - (A) passes through x = 0, y = 0
- (B) has a slee of +1

(C) is parallel to the *x*-axis

- (D) has a lope of -1
- Q.5 A foundry has a fixed daily cost of Rs 56,000 whenever it operates and a variable cost of Rs 800Q, where Q is the daily production in the es. What is the cost of production in Rs per tonne for a daily production of 100 tonne.

Q. 6 - Q. 10 carry two marks each.

Q.6 Find the odd one following group: ALRVX, EPVZB, ITZDF, OYEIK

(/ ALRVX

- (B) EPVZB
- (C) ITZDF
- (D) OYEIK

Q.7 Inuj, BhMa, Candan, Dilip, Eswar and Faisal live on different floors in a six-storeyed building ground floor is numbered 1, the floor above it 2, and so on). Anuj lives on an even-numbered floor Bholo does not live on an odd numbered floor. Chandan does not live on any of the floors below Faisal's floor. Dilip does not live on floor number 2. Eswar does not live on a floor immediately above or immediately below Bhola. Faisal lives three floors above Dilip. Which of the following floor-person combinations is correct?

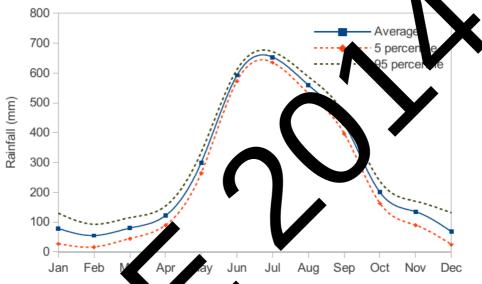
	Anuj	Bhola	Chandan	Dilip	Eswar	Faisal
(A)	6	2	5	1	3	4
(B)	2	6	5	1	3	4
(C)	4	2	6	3	1	5
(D)	2	4	6	1	3	5

GA 1/2

GATE 2014 SET- 1 General

Q.8 The smallest angle of a triangle is equal to two thirds of the smallest angle of a quadrilateral. The ratio between the angles of the quadrilateral is 3:4:5:6. The largest angle of the triangle is twice its smallest angle. What is the sum, in degrees, of the second largest angle of the triangle and the largest angle of the quadrilateral?

- Q.9 One percent of the people of country X are taller than 6 ft. Two percent of the people of country Y are taller than 6 ft. There are thrice as many people in country X as in country Y. Taking both countries together, what is the percentage of people taller than 6 ft?
 - (A) 3.0
- (B) 2.5
- (C) 1.5
- (D) 1.25
- Q.10 The monthly rainfall chart based on 50 years of rainfall in Agra is shown in the following figure. Which of the following are true? (*k* percentile is the value such that *k* percent of the data fall below that value)



- (i) On average, it rains muse in July it in Datember
- (ii) Every year, the are an of rainfall in fast is more than that in January
- (iii) July rainfall can be estimated with better confidence than February rainfall
- (iv) In August, there is at least 2 mm of rainfall
- (A) (i) and (ii)

(B) (i) and (iii)

(C) (ii) and (iii)

(D) (iii) and (iv)

END OF THE QUESTION PAPER

GA 2/2

Q. 1 - Q. 25 carry one mark each.

Q.1 For a real symmetric matrix [**A**], which of the following statements is true:

NPTEL Reference

- (A) The matrix is always diagonalizable and invertible.
- (B) The matrix is always invertible but not necessarily diagonalizable.
- (C) The matrix is always diagonalizable but not necessarily invertible.
- (D) The matrix is always neither diagonalizable nor invertible.
- Q.2 The series $S = \sum_{m=1}^{\infty} \frac{m!}{3^m} (x-2)^m$ converges for all x with $|x-2| \le R$ given by

NPTEL Reference

- (A) R = 0
- (B) R = 3
- (D) R=
- Q.3 The function given by $f(x) = \begin{cases} \sin(1/x), & x \neq 0 \\ 0, & x = 0 \end{cases}$ is

NPTEL Reference

- (A) Unbounded everywhere
- (B) Bounded and ious e
- **NPTEL Reference**

- (C) Bounded but not continuous at x=0
- ntial le everywhere (D) Continuous a d diffe.
- Q.4 Given the boundary-value problem $\frac{d}{dx} \left(x \frac{dy}{dx} \right) + ky = 0$, x < x < 1, with y(0) = y(1) = 0. Then the solutions of the boundary-value problem for k=1 (given by y_s and k=5 (given by y_s) satisfy:

(A)
$$\int_0^1 y_1 y_5 dx = 0$$

$$\int_0^1 \frac{dy}{dx} \frac{dy}{dx} dx = 0$$

NPTEL Reference

(C)
$$\int_{0}^{1} y_{1} y_{5} dx \neq 0$$

$$(D) \int_0^1 (y_1 y_5 + \frac{dy_1}{dx} \frac{dy_5}{dx}) dx = 0$$

- Q.5 The value of $I = \int_0^1 1000$ led by using Simpson's rule with 2 equally spaced intervals is,
 - (A) 200
- (C) 180
- (D) 208

NPTEL Reference

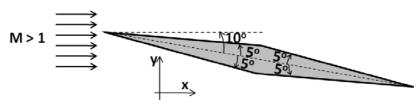
NPTEL Reference

- mord $oldsymbol{\varsigma}$ the designed lift coefficient and location of maximum Q.6 For a **NACA**5-dig camber along the cheed from the leading edge are denoted by C_L and X_m respectively. For **NACA 22018** foil, which combination of C_L and X_m given below are correct?

(B) $C_L = 0.12$ and $X_m = 0.2c$ (D) $C_L = 0.15$ and $X_m = 0.2c$

= 0.18c(C)**C**L

- d, supersonic flow over a diamond shaped airfoil, shown in the figure, which statement is Q.7 ct among the following?

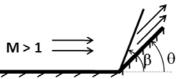


- (A) The airfoil will experience zero lift and positive drag force
- (B) The airfoil will experience positive lift and zero drag force
- (C) The airfoil will experience negative lift and zero drag force
- (D) The airfoil will experience positive lift and positive drag force

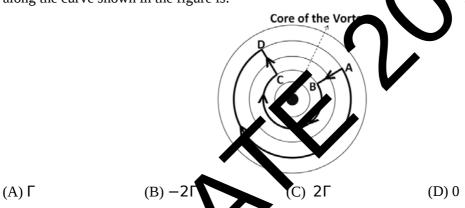
ΑE

Q.8 Consider supersonic flow near a corner (at an angle from the horizontal) with an attached oblique shock (at an angle β with horizontal) as shown in figure. If Mach number **M** decreases gradually from a high supersonic value, which of the following statements is correct?

NPTEL Reference

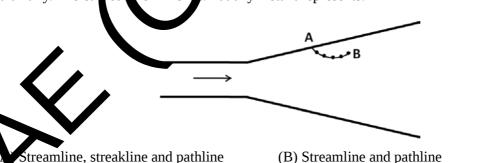


- (A) $\boldsymbol{\beta}$ will decrease if the shock is a weak shock
- (B) $\boldsymbol{\beta}$ will decrease if the shock is a strong shock
- (C) **B** will increase for both weak and strong shocks
- (D) **\beta**remains unchanged for both weak and strong shocks
- Q.9 The streamlines of a potential line vortex is concentric circles with respect to the vortex sentents shown in figure. Velocity along these streamlines, outside the core of the vortex case written as, $v_{\theta\theta} = \frac{\Gamma}{2\pi\pi}, \text{ where strength of the vortex is } \frac{\Gamma}{2\pi\pi} \text{ and } \textbf{\textit{r}} \text{ is radial direction.} \text{ The value of circulation along the curve shown in the figure is:} \\ \underline{\text{NPTEL Reference}}$



Q.10 To observe unsteady separated flow in a diverging channel, bubbles are injected at each 10*ms* interval at point A as shown in Aure These bubblesact as tracer particles and follow the flow faithfully. The curved are **Ab** shown at any instant represents:

NPTEL Reference



- Q.11 It is desired to measure the Young's modulus and the Poisson's ratio of a given homogeneous, isotropic material. A bar of length 20 *cm* and square cross-section (10 *mm* 10 *mm*) of this material is subjected to a tensile load of 40 *kN* Under this load, length increases to 20.1 *cm* while the cross-section reduces to 9.98 *mm* Young's modulus and Poisson's ratio of the material are:
 - (A) 80 *GPa*& 0.4 respectively

Only a pathline

(B) 40 *GPa*& -0.4 respectively

(D) Only a streakline

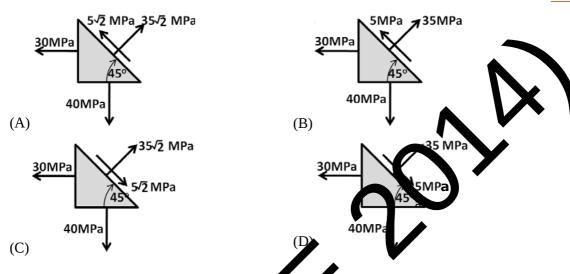
NPTEL Reference

NPTEL Reference

- (C) 80 *GPa*& -0.2 respectively
- (D) 40 *GPa*& 0.2 respectively

AE 2/11

- Q.12 In general, for any given solid subjected to arbitrary loading, which of the following statements is NPTEL Reference always:
 - (A) Volume does not vary with loading
- (B) Mass does not vary with loading
- (C) Density does not vary with loading
- (D) Volume, mass and density vary with loading
- Q.13 Which one of the following objects with inclined face at 45° , subjected to the given stresses, are in static equilibrium: NPTEL Reference



A damped single degree of freedom system damped natural frequency, $\omega \omega = 10 Hz$ is Q.14 subjected to sinusoidal external force. Po maximumfor the two frequencies of er is l 60.9469 rad/s and 64.7168 rad/s. The ing facto sociated with the vibrating system (in %) is

NPTEL Reference

NPTEL Reference

The boundary conditions for a roo ar cross-section, under torsional vibration, are changed Q.15 from fixed-free to fixedd. The ndamental natural frequency of the fixed-fixed rod is **k**times that of fixed-free rod. NPTEL Reference

(A) 1.5

b. High

vilian aircraft

ghter aircraft

(C) 2.0

(D) 0.5

engine (in right column) with the corresponding aircraft (in left column) for Q.16 Match the appropria ent performe se of the engine. most effi

transport aircraft spee

i. Ramjet

iii. Turbojet

iv. Turbofan

$$A = iv, b = iii, c = i, d = ii$$

nic aircraft

$$(C)$$
 $a - i$, $b - ii$, $c - iv$, $d - iii$

ii. Turboprop

(B)
$$a - ii$$
, $b - i$, $c - iii$, $d - iv$

(D)
$$a - ii$$
, $b - iv$, $c - iii$, $d - i$

Q.17 For a given fuel flow rate and thermal efficiency, the take-off thrust for a gas turbine engine burning aviation turbine fuel (considering fuel-air ratio f < 4) is

NPTEL Reference

- (A) Directly proportional to exhaust velocity
- (B) Inversely proportional to exhaust velocity
- (C) Independent of exhaust velocity
- (D) Directly proportional to the square of the exhaust velocity

ΑE 3/11 Q.18

II. The whirl or swirl component of absolute velocity at the entry of rotor and entry of stator are

For a fifty percent reaction axial compressor stage, following statements are given:

I. Velocity triangles at the entry and exit of the rotor are symmetrical

	same. Which of the follo	owing options are correc	rt?		NPTEL Reference			
	(A) Both I and II and (C) I is incorrect by	are correct statements out II is correct	(B) I is correct but (D) Both I and II a					
Q.19	A small rocket having a specific impulse of 200 s produces a total thrust of $98k$ N out of which $10k$ N is the pressure thrust. Considering the acceleration due to gravity to be $9.8 m/s$, the propellant mass flow rate in kg/s s							
	(A) 55.1	(B) 44.9	(C) 50	(D) 60.2				
Q.20	The thrust produced by a turbojet engine							
	(B) Decreases wit (C) Remains cons	n increasing compressor th increasing compressor tant with increasing con to and then decreases wit	r pressure ratio apressor pressure ratio	presseve ratio	NPTEL Reference			
Q.21	The moment coefficient measured about the centre of gravity and at jut ae odynamic centre of a given wing-body combination are 0.0065 and -0.0235 respectively. The aerodynamic centre lies 0.06 chord lengths ahead of the centre of gravity. The lift coefficient for this wing-body is							
Q.22	The vertical group	nd load factor on a static	onary aircraft parked in it	es hangar is:	NPTEL Reference			
Q.22	(A) 0	(B) -1	Not defined	(D) 1	NPTEL Reference			
Q.23		tionshould a glider oe o	peated to ensure minimum C_L/C_L	um sink rate?	NPTEL Reference			
	(C) Maximum C _D	${}^{\prime}\mathrm{C_{L}}^{3/2}$	(D) Minimum C_D	$C_{\rm L}^{}$				
Q.24	In most airplanes,	NPTEL Reference						
	(A) a step input to (C) a sinusoidal in		(B) a step input to (D) an impulse inp	the rudder out to the elevators				
Q.25	Considering R as the races of the moon, the ratio of the velocities of two spacecraft orbiting moon in circular orbit at altitudes R and 2 R above the surface of the moon is NPTEL Reference							
Q. 2	6 - 1 55 car	ry two marks o	each.					
Q.26	If $[A] = \begin{bmatrix} 3 & -3 \\ -3 & 4 \end{bmatrix}$. Then $\det(-[A]^2 + 7[$	<i>A</i>] – 3[/]) is		NPTEL Reference			
	(A) 0	(B) -324	(C) 324	(D) 6				

AE 4/11

Q.27

For the periodic function given by $f(x) = \begin{cases} -2, & -\pi < x < 0 \\ 2, & 0 < x < \pi \end{cases}$ with $f(x+2\pi) = f(x)$, using

Fourier series, the sum $S=1-\frac{1}{3}+\frac{1}{5}-\frac{1}{7}+\frac{1}{7}$ converges to

NPTEL Reference

(A) 1

(B) $\frac{\pi}{2}$ (C) $\frac{\pi}{4}$

(D) $\frac{\pi}{5}$

Q.28 Let Γ be the boundary of the closed circular region Agiven by $x^2 + y^2 \le 1$. Then $I = \int_{-1}^{1} (3x^3 - 9xy^2) ds$ (where *ds* means integration along the bounding curve) is

NPTEL Reference

(A) π

(B) $-\pi$

(C) 1

(D) 0

Solution to the boundary-value problem Q.29

$$-9\frac{d^2u}{dx^2} + u = 5x$$
, $0 < x < 3$ with $u(0) = 0$, $\frac{du}{dx} = 0$ is

NPTEL Reference

(A) $u(x) = \frac{15e}{1+e^2} (e^{-x/3} - e^{x/3}) + 5x$

(C) $u(x) = -\frac{15\sin(x/3)}{\cos(1)} + 5x$

Q.30 the problem $\frac{\partial^{p} u}{\partial t^{2}} + 2 \frac{\partial u}{\partial t} + u = 1, t > 0$ The Laplace transform L(u(t))=U(s), for the

 $\frac{du(0)}{du(0)} = 5 \text{ is given by:}$ with initial conditions U(0) = 0,

NPTEL Reference

(A) $\frac{6}{(s+1)^2}$

(C) $\frac{1-5s}{s(s+1)^2}$ (D) $\frac{5s^2+1}{s(s+1)^2}$

Q.31 For a steady, incorpres -dimensional flow, represented in Cartesian co-ordinates (xx,)ya student correctly write the equation of pathline of any arbitrary particle as $\frac{ddxx}{dddd} = aax$ and $\frac{ddyy}{dddd} = bbyy$ re constants having unit of (*second*). If value of *a* is 5, the value of *b* is ____

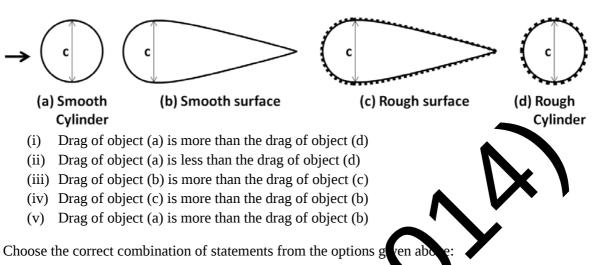
NPTEL Reference

5/11

ΑE

GATE 2014 AEROSPACE- AE

Q.32 Figures (a) - (d) below show four objects. Dimensions and surface conditions of the objects are shown in the respective figures. All four objects are placed independently in a steady, uniform flow of same velocity and the direction of flow is from left to right as shown in (a). The flow field can be considered as 2-D, viscous and incompressible. Following statements are made regarding the drag that these objects experience.



- (A) (i), (iii), (v)
- (B) (ii), (iv), (v)
- (C) (i), (iv) (v)
- (D (i), (iii)

Q.33 A student needs to find velocity across a stationary normal shoc. Ho measures density and pressure across the shock as shown in the figure bow. $1bar=10^{\circ}Pa$. (No shock table is needed for the calculations). The value of u_1 in m/s.

NPTEL Reference

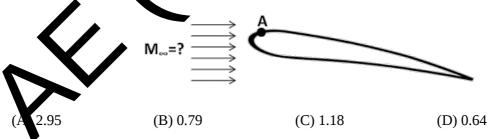
$$p_1 = 1 \text{ bar}$$

$$p_2 = 29 \text{ bar}$$

$$p_2 = 6 \text{kg/m}$$

$$u_1 \longrightarrow u_2$$

Q.34 For inviscid, compress ole flow, ast thin airfoil, shown in the figure, free-stream Mach number and pressure are denoted by MM and pp_{o} respectively. Ratio of pressure at point \mathbf{A} and pp_{o} is 0.8 and specific heat a tio is \mathbf{A} . If the Mach number at point \mathbf{A} is 1.0 and rest of the flow field is subsonic, the value of MM is



AE 6/11

GATE 2014 AEROSPACE- AE

Q.35 A student can measure free-stream velocity of a low-speed wind tunnel using a

- i. Pitot tube alone aligned with the flow direction.
- ii. Pitot tube aligned with the flow direction with static pressure measurement at an appropriate position on the tunnel wall.
- iii. Pitot tube aligned with the flow direction along with barometer pressure reading of the outside ambient.
- iv. Pitot static tube alone aligned with the flow direction.

Considering the above statements, which of the following options is correct?

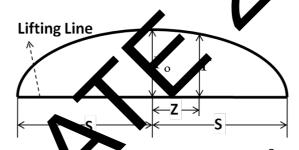
- (A) (i) only
- (B) (i) & (ii)
- (C) (ii) & (iv)
- (D) (i), (iii) & (iv)

Q.36 Induced velocity W at a point $zz = \sqrt{3}z$ along the lifting line can be calculated using the formula $w(z) = -\frac{1}{4\pi} \int_{-ss}^{ss} \frac{dd\Gamma}{ddz} \frac{1}{zz - z} ddz$

Given $\frac{\Gamma^2}{\Gamma_{oo}^2} + \frac{zz^2}{sz^2} = 1$, where zz, $\int_0^z and S$ are given in figure below.

For the above semi-elliptic distribution of circulation, Γ , the downwash velocity at any point $z_{\bar{1}}$, for symmetric flight can be obtained as, $w(z_{\bar{1}}) = \frac{\Gamma_{oo}}{4\pi\pi} [\pi\pi + 1]$, where $\Pi = z \int_{-ss}^{ss} \frac{ddzz}{s(z-zz)}$

Which of the following options is correct if the induced dog is $p(given \int_{-ss}^{s} \sqrt{1 - \frac{zz^2}{ss^2}} ddzz = \frac{\pi\pi s}{2})^s$



(A)/=
$$\theta$$
and D_{t} = $\frac{8\varrho\varrho \frac{2}{6\rho}}{\pi\pi}$

$$=\frac{\pi\pi\varrho_{0}^{2}}{\pi^{2}}$$

(B)/=
$$I$$
 and D_{P} = $\frac{8\varrho\varrho}{\pi\pi}$ (D)/= I and D_{P} = $\frac{\pi\pi\varrho}{8}$ (D)/= I

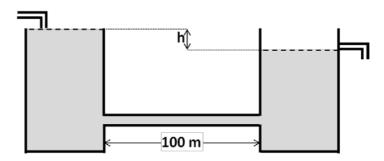
Q.37 Two overflowing water reservoirs are connected with a100*m*long pipe of circular cross-section (of radius, *R* = 0.02*m*), such that height difference *h* remains constant as shown in the figure below. The certaining velocity in the pipe is 10*m*/*s*The velocity profile inside the pipe over the entire

length $A = \frac{RR \, dddh}{\mu u \, dx} - \frac{\pi \pi}{RR}$ where, $\frac{dddd}{ddx}$ is a constant pressure gradient along the pipe length,

The peasure from the left end of the pipe along its central axis and r is radial location inside the pipe and espect to its axis. (*Given dat* Density and kinematic viscosity of water are 1000kg/m at $1x10^{-6}m/s$ espectively; acceleration due to gravity is 10m/s).

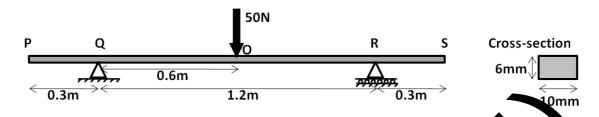
If \mathbf{l} other losses except the frictional losses at the pipe wall are neglected, the value of \mathbf{h} in meter is \mathbf{l} .

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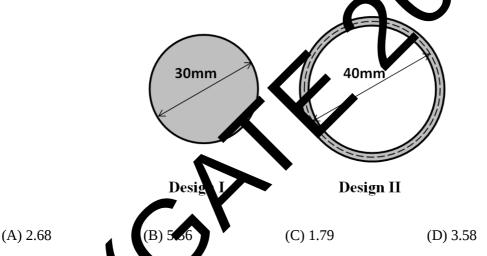
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Q.38 A 1.8 *m*long steel beam of rectangular cross section (10*mm* 6 *mm*) is simply supported with a length of 1.2 *m* between the supports and an overhang of 0.3 *m* on either side. Young's modulus for the material of the beam is 200 *GPa*. For a 50 *N* load applied at the center of the beam, magnitude of the slope of the beam at tip *S* is ______. NPTEL Reference

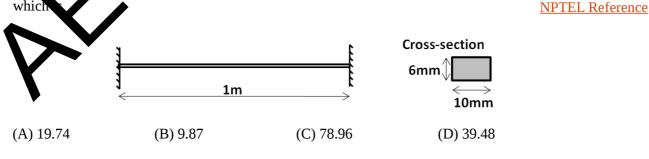


Q.39 There are 2 designs proposed for a shaft of length *l*, with a torque carrying capacity of **Design is a solid circular cross-section shaft of diameter 30mmDesign is a thin-valled circular shaft of average diameter 40mmThickness of the wall in *Design** last to be determined such that maximum shear stress is the same in both the designs for the given a rate **T**(so hardsame material can be used for manufacturing both the shafts). Ratio of mass of haft using *Design** like*

NPTEL Reference**



Q.40 A structural member of recognilar cross-section 10mm 6mm and length 1m is made of steel (Young's modulus is 100 GPa and coefficient of thermal expansion is 12×10^6 / ^{o}CC It is rigidly fixed at 1 of the ends and then subjected to a gradual increase in temperature. Ignoring the three dimersional effects, the structural member will buckle if the temperature is increased by $\Delta TT^{o}CC$



Q.41 A gas cylinder (closed thin-walled cylindrical pressure vessel) of diameter 30*cm* and wall thickness 1*mm* subjected to a design maximum internal pressure of 5*bar*(0.5*MPa*). The material used for manufacturing this cylinder has a failure stress of 260*MPa*. Assuming von Mises failure criterion, the factor of safety (with respect to maximum allowable stress) for this cylinder is

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(A) 2.8

(B) 2.0

(C) 6.9

(D) 4.0

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Q.42 A cantilevered beam is subjected to a parabolic distribution of shear traction at the right edge while the top and bottom surfaces are traction free. To solve this problem, following Airy's stress function is proposed: $\phi \phi = \alpha G y + 2 G \partial y + G \partial x \partial y + G \partial x \partial y + G \partial x \partial y$ This is an admissible Airy's function that would satisfy the bi-harmonic equation as well as the boundary conditions if and only

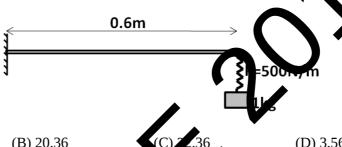


(A)
$$CC = QCC = \lambda \lambda CC = QCC = \frac{\lambda \lambda}{3dd}$$

(B)
$$CC = \lambda \lambda CC = \frac{\lambda \lambda}{3\sqrt{d}}$$
, $CC = 0$, $CC = 0$

(A)
$$CC = 0$$
, $CC = \lambda \lambda CC = 0$, $CC = \frac{\lambda \lambda}{3dd}$ (B) $CC = \lambda \lambda CC = \frac{\lambda \lambda}{3dd}$, $CC = 0$, $CC = 0$ (C) $CC = 0$, $CC = \lambda \lambda CC = \frac{\lambda \lambda}{3dd}$, $CC = 0$, $CC = 0$

1kg mass is hanging from a spring of stiffness 500*N*/*m*attached to a massless,sym ne Q.43 length 0.6m moment of inertia about the bending axis $I = 8.33 \times 10^{6} \text{ m/h}$ and Yd *E*=210*GPa* as shown in the figure. The fundamental natural frequency (i **⊋**d/**y**of



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- (A) 3.24
- (B) 20.36
- (D) 3.56
- Q.44 A single degree of freedom system is al (first cycle) amplitude of 5*cm*The viscous damping factor associated with the brating system is 2% Vibration amplitude of the fifth cycle (in cm) is **NPTEL Reference**

- (A) 1.65
- (B) 4.41
- (C) 2.67
- (D) 3.02
- Q.45 A cruise missile with a njet ingine is flying at Mach 4.0 at an altitude where the ambient temperature is 100K. of specific heatsy = 1.4 and specific gas constant R = $287 \frac{l}{kgK}$ onside rati combustion chamber is equal to 2310K, the speed of the exhaust If the stagnation temperature ure in gases (in *m/***\$**is

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Q.46 oine engine is operating under the following conditions:

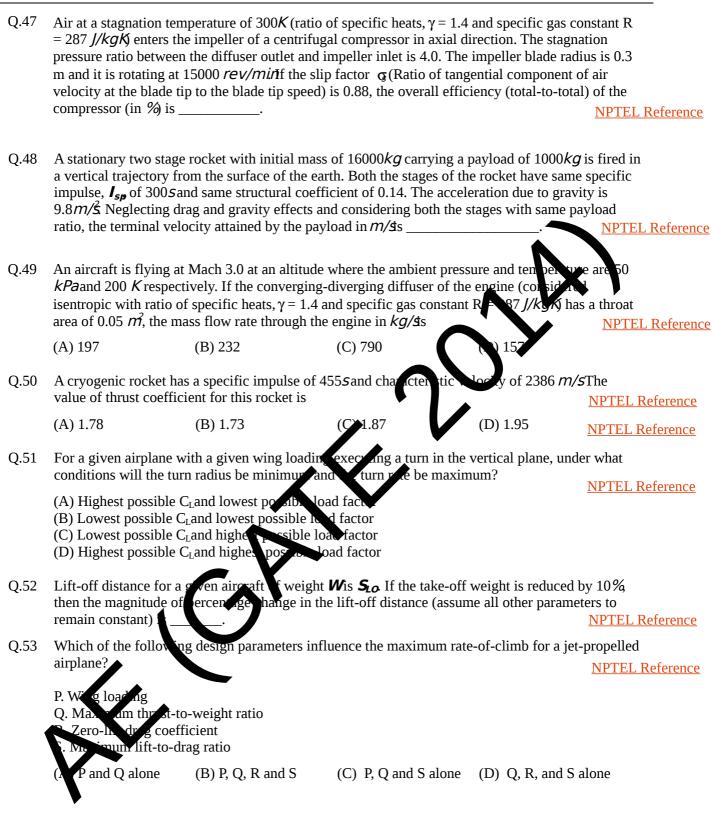
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temp ature at turbine inlet 1350 *K* Stagna 10 *bar* sure at the turbine inlet gnation perature at turbine exit 800 K city at turbine exit 200 *m/s* l-to-total efficiency of turbine 0.96 γ (ratio of specific heats) 1.33

1.147 *kJ/kgK* C_P (specific heat at constant pressure)

The stagnation pressure (in bar) in the nozzle (considering isentropic nozzle) is equal to _____

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Q.54 Consider the following four statements regarding aircraft longitudinal stability:

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- P. $C_{M,cg}$ at zero lift must be positive
- Q. $\partial C_{M,c} \partial C_{M}$ must be negative (C_{M} is absolute angle of attack)
- R. $C_{M,cg}$ at zero lift must be negative
- S. Slope of C_L versus α_a must be negative

Which of the following combination is the necessary criterion for stick fixed longitudinal balance and static stability?

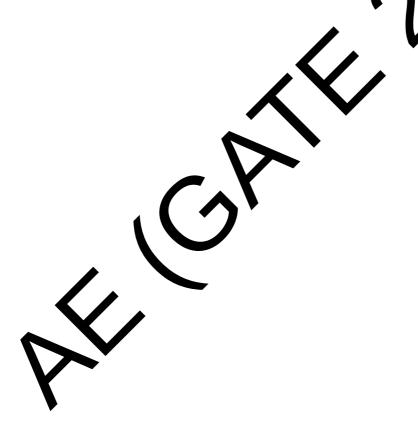
- (A) Q and R only
- (B) Q, R, and S only
- (C) P and Q only
- (D) Q and S only
- Q.55 Data for a light, single-engine, propeller driven aircraft in steady level flight at sea-level as follows: velocity $V_{\infty} = 40 \, m/\varsigma$ weight $W = 13000 \, N$, lift coefficient $C_L = 0.65$, drag c efficient $C_D = 0.025 + 0.04 \, C_L^2$ and power available $P_{av} = 100,000 \, J/s$. The rate of climb possible or his aircraft under the given conditions (in m/ς is
 - (A) 7.20
- (B) 5.11
- (C) 6.32

 $\langle m \rangle \chi_{2}$

NPTEL Reference

NPTEL Reference

END OF THE QUESTION PAREL



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